



Thermoplastics Development for Exploration Applications (TDEA) Project

Programmatic and Technical Overview

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Thermoplastic Development for Exploration Applications (TDEA) Project Overview

Goal:

Advance NASA's thermoplastic composites capabilities by developing structurally efficient joining solutions for large-scale space structures and applications to support NASA's future exploration missions.

Objectives:

1. Assess the current capability for design, analysis, and manufacture of commercially available thermoplastic material systems for large-scale space structures and on-orbit applications.
2. Evaluate thermoplastic composite material systems for large-scale space structures and on-orbit applications.
3. Develop NASA's in-house capabilities in the area of thermoplastic composites processing and manufacturing.
4. Develop and understand advanced thermoplastic joining technique(s) relevant to space environments and applicable to unitized and/or reconfigurable composite structures.
5. Advance structural analysis capabilities for the design and analysis including failure prediction of thermoplastic composites including joints.

Team:

MSFC (lead),
GSFC,
GRC,
LaRC

Budget:

\$10.5M, FY22-24

Game Changing
Development Program

Why Thermoplastic Composites?

Qualitative comparison of thermoplastic composites (TPC) and thermoset composites (TSC)

Advantages:

- Reduced cycle time
- **Processing by remelting**
- Automated assembly (robotic welding)
- **Higher fracture toughness**
- **Ambient storage, no shelf life**
- **Welded joints with no material interface**
- **Processing that enables unitization**
- **Minimal outgassing**
- Low moisture uptake

Disadvantages:

- **Higher processing temperature and pressure required**
- **Higher residual stresses (more difficult dimensional control)**
- Higher raw material cost
- Structural and chemical properties sensitive to crystallinity
- Higher melted viscosity
- Crystallinity may change over lifecycle
- Lower TRL

Bolded characteristics especially relevant for space applications

Selected property comparison

Property	TPC T700/LM PAEK ^b	TSC IM7/8552 ^a
Density, ρ	1.6 g/cm ³	1.6 g/cm ³
Glass transition temperature ^c , T_g	147°C	200°C
Transverse stiffness, E_{22}	9.1 GPa	8.7 GPa
Transverse tensile strength ^d , Y_T	94 MPa	80 MPa
Interlaminar toughness, G_{1c}	2.1 kJ/m ²	0.24

^aWanthal, et al., ASC Conference, 2017

^bNCAMP database, report CAM-RP-2019-036 RevA

^cNeat resin, dry, from manufacturer's datasheet

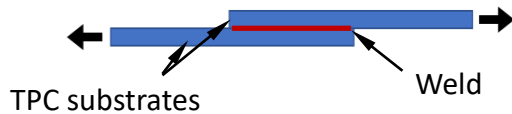
^dASTM D3039

Recommendations: High Priority TPC Developments for Space Applications

Recommendation	Objectives	TDEA Contribution
1 Launch vehicle structure trade study: TSC vs. TPC	<ul style="list-style-type: none">Quantify the weight and costs savings for TPC compared to SoA TSC baseline.Characterize the applicability of TPCs for launch vehicle cryotanks.	Preliminary assessment of weight savings with postbuckled TPC design.
2 TPC material qual. for space applications	<ul style="list-style-type: none">Generate space-relevant material properties for candidate TPC materials to support future trade studies and PDR level analysis.	Laminate mechanical performance: AS4/PEEK, AS4/PPS, T700/LM PAEK, AS4/PEI (additional mat'ls and tests TDB)
3 Demonstrate capability for TPC welded joints (flat coupons)	<ul style="list-style-type: none">Confidence in TPC welded joint design, sizing, and fabrication at the coupon scale.Analysis tools and relevant experience to assess welding methods for terrestrial and in-space assembly (ISA).Characterization of joint behavior in off-nominal environments.	Welded joint pathfinder study
4 Evaluate cost & benefits for low-production-volume TPC parts	<ul style="list-style-type: none">Demonstrate cost-effective materials, process, and structural performance for specialized space application structures (currently made with TSC or metals).	Roman Space Telescope (RST) Deployable Aperture Cover (DAC) 'conceptual' TPC support beam
5 De-risk TPC manufacturing, design, analysis, and inspection technology for in-space construction	<ul style="list-style-type: none">Ground test data for evaluation of ISA with TPCs.Validated process monitoring and modeling for high-confidence in-space operations.Tool-less TPC AFP technologies suitable for in-space construction of pressure vessels.	Element/subcomponent test and analysis for ISA using TPCs

TDEA Technology Roadmap

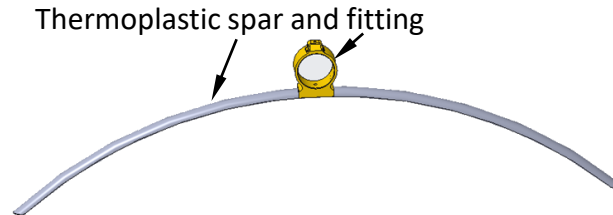
Welded joint pathfinder *Foundational developments*



- Coupon scale, **single welded interface**
- Develop analysis approaches for welded joints trade studies
- Fabricate and test joints with nominal conditions, and conditions relevant to in-space applications (e.g., off-nominal thermal history, lunar dust contamination)

- Seek in-house capabilities for TPC welding relevant to space applications
- Simple generic configuration to leverage existing data

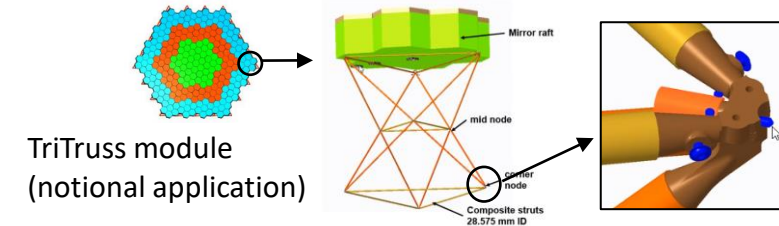
Roman Space Telescope (RST) support beam *Confidence building application*



- Moderate-scale, **single spar and fitting**
- Terrestrially manufactured and assembled spacecraft structure
- Evaluate approaches that reduce cost through reduced part count and complexity compared with the RST baseline design

- Knowledge gap in application of TPCs to low-production-volume applications
- How can processes unique to TPCs (e.g., welding, additive) enable better designs?

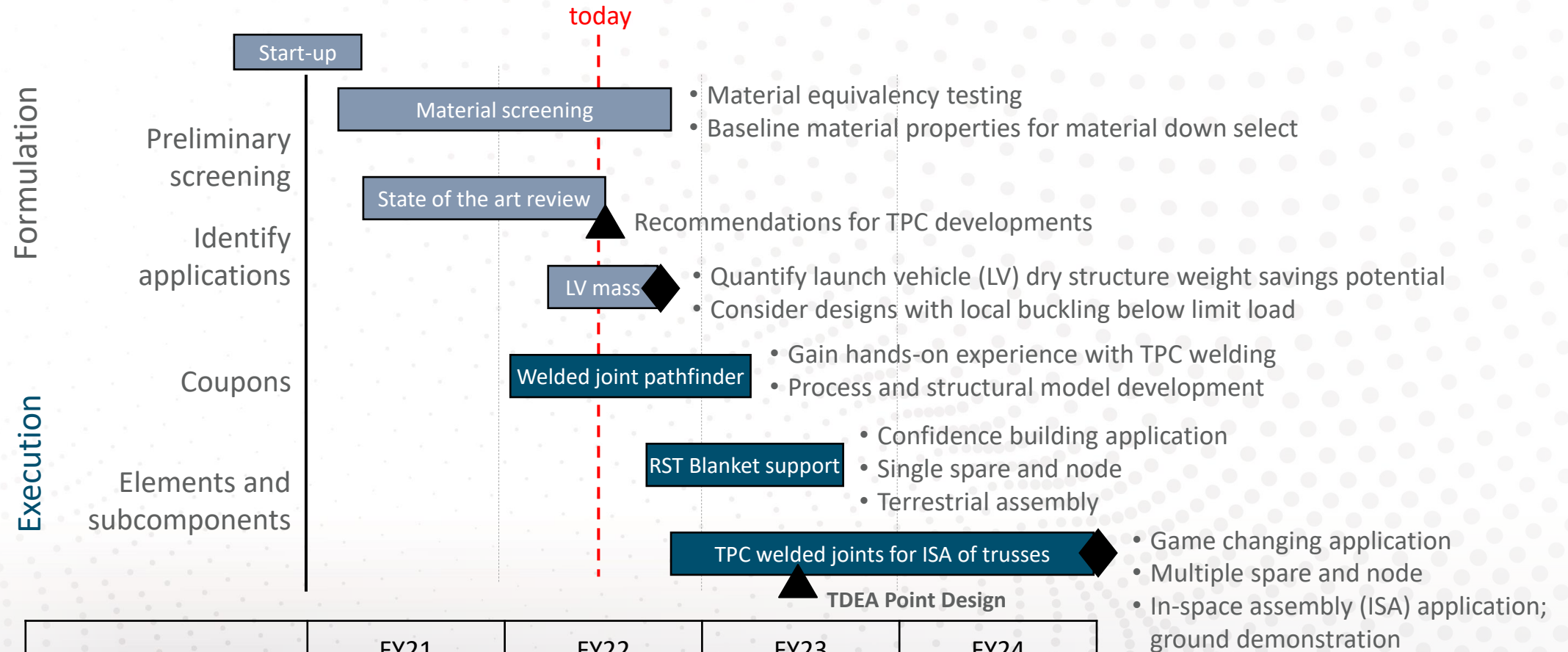
Game Changing application *Developments for in-space TPC truss assembly*



- Large-scale, **multiple spars and fittings**
- Terrestrially manufacturing and in-space assembly (ISA)
- Identify relevant welded joint design requirements, and establish a TDEA Point Design
- Demonstrate through the building block approach

- Challenges remain for TPCs use in ISA/OSAM
- Develop deep understanding of considerations for TPC joining in-space

TDEA Planned Activities and Schedule



	FY21	FY22	FY23	FY24
FY21 Budget Request	Worked as part of CTE project	\$2M	\$3.5M	\$5M
Civil servant FTE		9	10	10